#### **F-35 Class Hovercraft Propulsion**

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# Goal

 Determine whether the University of Alabama Hoverteam F-35 Class hovercraft propulsion system should use a non-ducted propeller, a ducted propeller, a non-ducted fan, or a ducted fan

• Determine the diameter of the propeller/fan

## Overview

- How does a hovercraft work
- Hovercraft racing classes
- Hoverteam craft design
- Factors/Configurations to consider
- Non-Ducted Thrust Calculations
- Ducted Thrust Calculations
- Weight Calculations
- Selection of propulsion system

#### How does a hovercraft work?



#### How does a hovercraft work?



## Hovercraft Racing Classes

- Formula 1 no upper limit on size or # of engines
- Formula 2 no limit on # of engines, upper limit on engine

size

- Formula S single engine, fan, duct, no engine size limit
- Formula 50 single engine, fan, duct, 50 HP
- Formula 35 no limit on # of engines, total of 35 HP



- Formula 35 class
- 27 HP thrust engine
- 6.5 HP lift engine

#### Factors to Consider

- Manufacturer availability
- Thrust/Weight would like this to be high
- Safety HCA guidelines
- Ability to change from fan to propeller
- Noise HCA guidelines

## **Configurations Considered**

- Non-ducted propeller
  - 2 blades
  - 3.5 ft.
  - 4 ft.
- Ducted propeller
  - 2 blades
  - 3.5 ft.
  - 4 ft.
- Non-ducted fan
  - 5 blades
  - 3.5 ft.
  - 3.75 ft.

#### • Ducted fan

- 5 blades
- 3.5 ft.
- 3.75 ft.





## **Thrust Calculation**

- Momentum theory will be used
- Constants

$$-P = 26.5 hp = 14575 \frac{ft * lb}{s}$$
$$-\rho = 0.0023769 \frac{slugs}{ft^3}$$

- Assumptions made:
  - Steady flow
  - Incompressible flow
  - Neglect rotation imparted to flow
  - Air at standard temperature and pressure

#### **Momentum Theory**



#### Non-ducted Thrust Calculation

• Determine  $V_1$  using propeller efficiency

- Ideal propeller efficiency  $(n_{pr_{ideal}})$  is about 0.8

$$-n_{pr_{ideal}} = \frac{V_0}{V_1}$$
  

$$-n_{pr} = 0.85 * n_{pr_{ideal}} = 0.85 * 0.8 = 0.68$$
  

$$-0.68 = 0.85 * n_{pr_{ideal}} = 0.85 * \frac{V_0}{V_1}$$
  

$$-V_1 = 1.25 * V_0$$

#### Non-ducted Thrust Calculation

• Find V<sub>3</sub> using power equation

$$-P = \dot{m} \left( \frac{V_3^2}{2} - \frac{V_0^2}{2} \right) = \rho V_1 A_{disk} \left( \frac{V_3^2}{2} - \frac{V_0^2}{2} \right)$$
  
- Solve for  $V_3 = \sqrt{\frac{2P}{\rho V_1 A_{disk}} + V_0^2}$ 

• Solve for thrust

$$-T = \dot{m}(V_3 - V_0) = \rho V_1 A_{disk}(V_3 - V_0)$$

## Non-ducted Thrust Calculation

• Calculations are performed for values of  $V_0$  ranging from 5 mph to 60 mph

60 mph is the approximate top speed of the craft

• Unlikely that the craft will be going this fast during a race

#### Non-Ducted Thrust Comparison

Thrust Produced vs. Freestream Velocity Non-Ducted Fan/Prop



# Why is a duct helpful?

 Helps prevent pressure tip vortices from forming

Decreases noise due to reduction of tip vortices

 Essentially has the same effect as if you were increasing the diameter of the prop/fan

## Why is a duct helpful?

• Decreases turbulence at the blade tip



## **Ducted Thrust Calculation**

• For this propeller/fan size range, a duct **can** cause an increase in thrust of up to 25%

• Generally, a hovercraft thrust duct will increase the total thrust output by 10-15%

• For calculations, assume a 10% increase in total thrust output

## **Ducted Thrust Calculation**

• Assumptions

Duct does not converge or diverge



## **Ducted Thrust Calculation**

- Multiply thrust values found for non-ducted propeller/fan by 1.1 to show a 10% increase in total thrust output
  - Assumptions (incompressible, steady, etc.) still apply

$$-T_{ducted} = 1.10 * T_{non-ducted}$$

#### **Ducted Thrust Comparison**

**Thrust Produced vs. Free Stream Velocity** Non-Ducted and Ducted Fan/Prop 180 160 140 Non-Ducted 3.5 ft Thrust (lbf) Non-Ducted 3.75 ft 120 Non-Ducted 4 ft Ducted 3.5 ft 100 Ducted 3.75 ft Ducted 4 ft 80 60 20 30 70 10 0 40 50 60 Free Stream Velocity, V<sub>0</sub> (mph)

## **Propeller Weight Calculation**

- 4 ft. propeller weighs 5.2 lbs.  $\rightarrow$  1.3 lbs/ft
  - Propeller in the hovercraft lab was weighed

-W = 1.3 \* Diameter

Diameter (ft)	Fan/Prop	Total Weight (lbs)	
3.5	Propeller	4.55	
4	Propeller	5.20	

## Fan Weight Calculation

- Fan hub weighs about 3.5 lbs and is 11" in diameter
- Each blade for a 45" fan weighs 0.9 lbs and is 17" long → 0.635 lbs/ft
- A five bladed fan will be used for calculations

$$-W = 3.5 + 0.635 * 5 * \left(\frac{Diameter - 0.916}{2}\right)$$

Diameter (ft)	Fan/Prop	Total Weight (lbs)	
3.5	Fan	7.60	
3.75	Fan	8.00	

## **Duct Weight Calculation**

- Mid range Styrofoam density is 2.275 lbs/ft<sup>3</sup>
- Duct assumed perfectly cylindrical, 18" in height, and 3" thick

$$-W = 2.275 * (\pi r_{outer}^{2}h - \pi r_{inner}^{2}h)$$

Diameter (ft)	Weight(lbs)
3.5	8.71
3.75	9.38
4	10.05

## **Total Assembly Weight**

Diameter (ft)	Fan/Prop	Duct/Non	Total Weight (lbs)
3.5	Propeller	Non	4.55
4	Propeller	Non	5.20
3.5	Fan	Non	7.60
3.75	Fan	Non	8.00
3.5	Propeller	Duct	13.26
4	Propeller	Duct	15.25
3.5	Fan	Duct	16.31
3.75	Fan	Duct	17.38

#### **Final Decision**

HCA mandates that a duct be installed on the craft for safety reasons

- For this reason, non-ducted propellers and fans must be eliminated from consideration
- Noise will be reduced with the addition of the duct

## **Thrust/Weight Comparison**

Thrust/Weight vs. Free Stream Velocity Ducted Fan/Prop



#### **Final Decision**

Factor	Best Choice	
Thrust/Weight	3.5 ft. Ducted Propeller	
Safety	Any Ducted Propeller	
Interchangeable	3.5 ft. Ducted Fan/Propeller	
Noise	4 ft. Ducted Propeller	

#### Best Choice: 3.5 foot ducted propeller